# AIR FORCE APPLICATION OF ADVANCED MAGNETIC MATERIALS

## DARPA WORKSHOP ON METAMATERIALS GREENBELT, MARYLAND



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### **Outline**





 Unmanned Combat Air Vehicle (UCAV)

Directed EnergyWeapons

Summary



## **Propulsion Directorate**







### **Power Division Four Focus Areas**



**ELECTRICAL COMPONENTS** 

**CAPACITORS** 





**FOUNDATIONAL** 

**POWER** 

**COMPONENTS** 

RESEARCH

AND

DEVELOPMENT



**GENERATORS** 



**AUXILIARY POWER UNITS** 



**ELECTRO-CHEMISTRY** 



**THERMAL MANAGEMENT** 





**SUPER-**CONDUCTIVITY



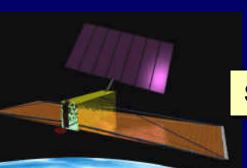
**PLASMA PHYSICS** 

**MEA VISIONARY CONCEPT** Internal Engine Electric Electric Driven. Starter / Generator **Environmental and** Integrated Power Unit **Engine Controls** Solid State Power Electric Aircraft Controllers Utility Functions Solid State Remote Terminal Solid-State Electrical **Electric Driven** Distribution System Flight Actuators Electric Electric Actuated Brakes

**AIRCRAFT** 



**WEAPONRY** 



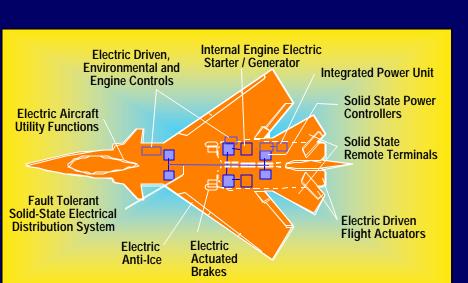
**SPACECRAFT** 



### **More Electric Aircraft**



#### THE VISION



- ALL-ELECTRIC AIRCRAFT (NO HYDRAULICS OR BLEED AIR PNEUMATICS)
- ELIMINATION OF ACCESSORY DRIVE GEARBOX (REDUCED FRONTAL AREA)

ENABLES MISSION AVAILABLE POWER FOR LETHAL AIRBORNE DIRECTED ENERGY WEAPON

#### THE IMPACT



- REDUCED LCC
- DRAMATIC IMPROVEMENT IN R, M, & S
- REDUCED DEPLOYMENT FOOTPRINT AND MANPOWER
- INCREASED SORTIE GENERATION RATE

SAVINGS IN \$B& WITH IMPROVED WARFIGHTING

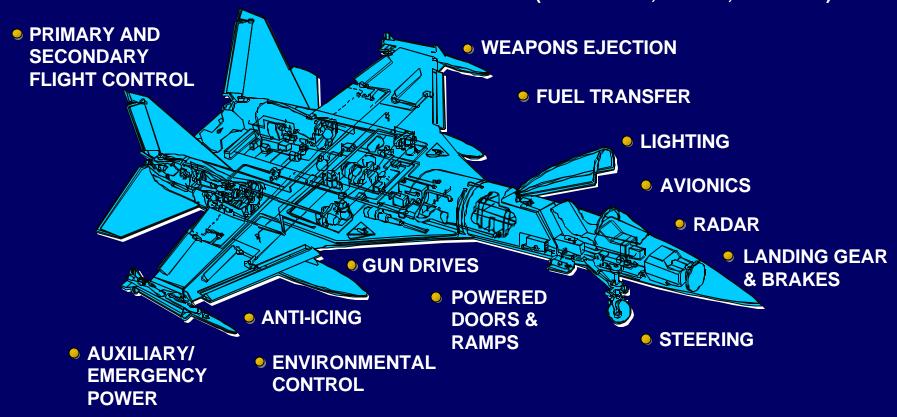
1992 LMTAS, 1995 VITPS, & 1998 J/IST STUDIES



## Conventional Non-Propulsive Aircraft Power



PROPULSION SYSTEM EXTERNALS (ACTUATION, PUMPS, STARTER)



A COMPLEX, HEAVY, HIGH MAINTENANCE HYBRID OF MECHANICAL, HYDRAULIC, PNEUMATIC AND ELECTRICAL SUBSYSTEMS



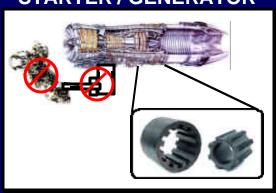
## **Key MEA Technologies**



 TRANSITION TO ELECTRICALLY DRIVEN CONCEPTS DEPENDS ON DEMONSTRATION OF FOUR KEY SUBSYSTEM TECHNOLOGIES

## INTERNAL INTEGRAL STARTER / GENERATOR

- EliminatesGearbox
- More Electric Engine FY05-07



## FAULT-TOLERANT DISTRIBUTION



- Redundant/ Reliable Electric Power
- Flight Worthy Hardware FY03-05

#### **INTEGRATED POWER UNIT**

- Eliminates GSE
- 6.3 Program Complete FY00-02



#### **ELECTRIC ACTUATION**



- Eliminates Centralized Hydraulics
- High HP, Low Cost Stabilator Hardware FY02-04

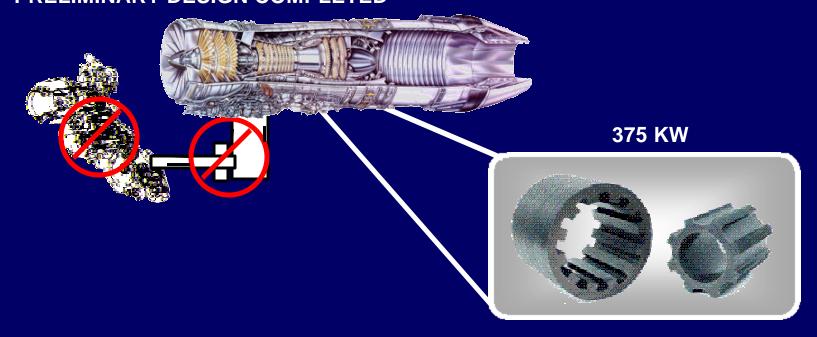
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# Revolutionary Internal Engine Starter / Generator



 INTERNAL F110 ENGINE ELECTRICAL STARTER AND GENERATOR PRELIMINARY DESIGN COMPLETED



- ELIMINATES ENGINE GEARBOXES HUGE REDUCTION IN PARTS
- REPLACE WITH SINGLE COMPONENT SOLID ROTOR ELECTRIC MACHINE
- COMBINES STARTER AND GENERATOR FUNCTIONS
- REDUCES FRONTAL AREA



## Internal Starter / Generator (IS/G)



#### **Aircraft Main Power**

 IS/G benefits both manned & unmanned aircraft and is part of IHPTET Phase III

Work with outside organizations for magnetic materials research (DARPA & AFOSR)

Collaborate with IHPTET organizations for engine integration and testing

Develop related technologies necessary to meet Gen II goals







- 550-600°C Mag Materials
- 400°C Insulation
- 1.5kW/#
- >15,000 Hrs MTBF



## **Integrated Power Unit (IPU)**



#### **Auxiliary/Emergency Power**

IPU has significant payoff for reduced supportability



6.2



6.3



**Dual-Use** 

- Component development (SR starter/generator & high speed magnetic bearings)
- Model integration of rotor dynamics, electrical & controls
- Integrate SR machine and magnetic bearings on one rotor
- Couple SRM/magnetic bearings with existing turbomachine
- Improve controls, electronics and SR machine for reliability & life



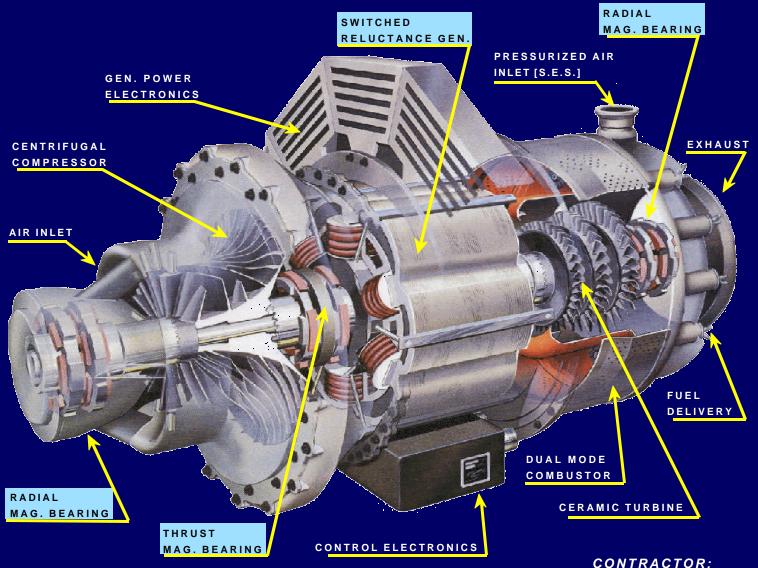


- 5000 Hrs MTBF
- 400 HP/ft3
- 550-600°C Mag Materials



## **More Electric Aircraft Integrated Power Unit**





ALLIED SIGNAL / AVCON



# Switched Reluctance Motor (SRM) Design Issues



#### **SRM Cooling Requirements**

- Heat transfer coefficients
- Windage losses
- Magnetic losses
- Axial conduction

#### **Stator Winding**

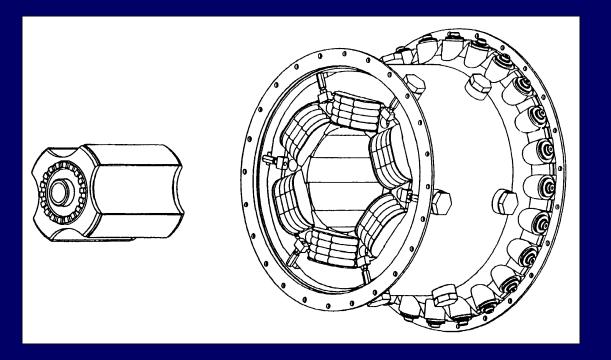
- Temperatures
- Frequency

#### **Rotor Integrity**

- Temperature distribution
- Tip speed
- Material properties
- Life prediction
- Laminate Stiffness

#### **Stator Integrity**

- Temperature distribution
- Impregnation





## MEA IPU SRM Subsystem Test Rig (Rotor & Stator)

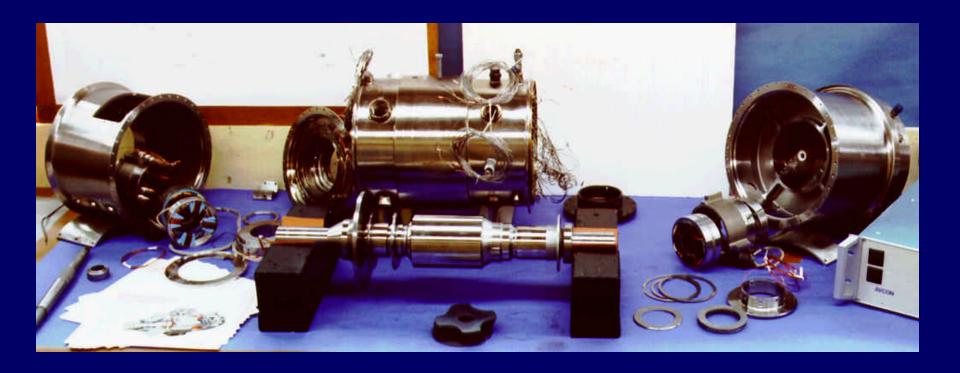






## MEA IPU Rotor / Bearing Subsystem Test Rig/Rig Disassembled







## **Magnetic Bearings**



- Electromagnetic biased bearing
  - Heavy/high power consumption
  - High temperature capable



- Permanent magnet biased bearing
  - Lower weight/ lower power
  - Smaller than electromagnetic bearings
  - Temperature limited





## **UAV Technologies**



Next Generation Swept & Splittered Fans

Integral Starter-Generator

Fully Integrated IR/RCS/Acoustic Signature Technologies

High Temperature Lubricants & Bearings

or

**Magnetic Bearings** 

Active Stall Control

Thermal
Management
Controls

Health
Management
Control System

Advanced Organic Matrix Composites

Advanced Manufacturing Processes

Fluidic Area & Vectoring Nozzle Control



## Unmanned Combat Air Vehicle (UCAV)



- Integral Part of US Force Structure by 2010
  - Clean Sheet Designs
  - Increased Tactical Deterrence
  - Reduced Life-Cycle Costs (50%-80%)

 High Speed Starter/Generator Systems are Likely (Internal to main propulsion engine)

- Totally Lubeless/Oilless Aircraft for Improved A/R/M/S and Long Term Storability
  - Electromechanical Actuation
  - Air-Only Cooling
  - Electric Secondary Power System

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## Power Technologies Enable Directed Energy ...



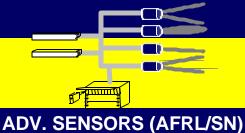


#### OTHER DIRECTORATE











REVOLUTIONARY SENSORS



IPU

IIS/G







**HPM UCAV** 



HTSC GENERATORS

**PMAD** 

ADVANCED TURBINES



ADVANCED LASER

(AFRL/DE)



HYPER-SONICS

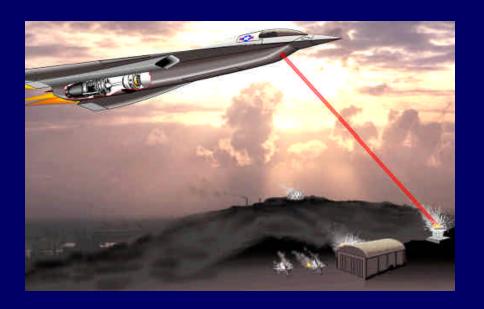
**LETHAL** 

NON-LETHAL



## **Directed Energy Weapons (DEW)**





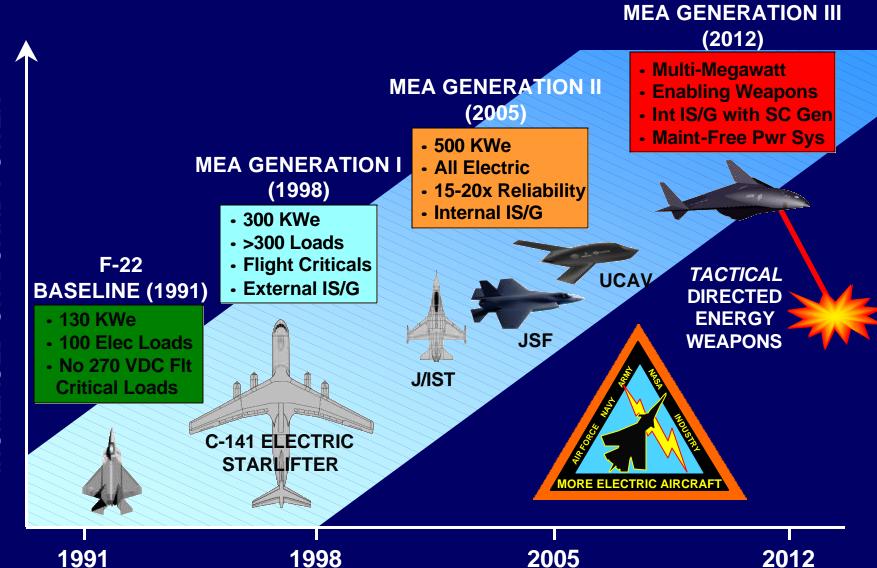
- Ground, Sea, and Air Platforms
- Projected to Totally Eliminate the need for Conventional Weapons by 2030
- Require High Power Light-Weight Energy Sources

- Critical Technology Areas Include:
  - Increased Electric Power Generation and Storage
  - Self Starting Capabilities
  - High Speed Magnetic Bearings
  - Air Only Cooling



## **Aircraft Power Capabilities**







### **Bottom Line**



#### ROTOR APPLICATION (SOFT MAGNETIC MATERIAL)

OPERATING TEMPERATURE

MACHINE CORE LOSS

600°C For 5000 Hours

700 MPa @ 500°C; 825 MPa 300°C

700 MPa @ 500°C; 825 MPa 300°C

2 x10-6 hr-1 for 5000 hours

(@ 550°C & 600 MPa)

RESISTIVITY

40 --> 60 micro-ohm-cm @ RT

#### BEARING APPLICATION (HARD MAGNETIC MATERIAL)

OPERATING TEMPERATURE 450°C for 5000 Hours

ENERGY PRODUCT 30MGOe @ 450°C

RESISTIVITY 60 --> 120 micro-ohm-cm @ RT



### **Summary**





Today systems are typically engineered around magnetic material temperature limitations.

Higher operating temperature magnets enable simpler, more efficient designs for many commercial and military applications.

As commercial and military electromechanical applications continue to evolve power system requirements continue to become more extreme.

New magnetic material systems are crucial to enable advanced power generation, distribution, and utilization systems.

